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APPLICATION NO.	FILING DATE	FIRST NAMED INVENTOR	ATTORNEY DOCKET NO.	CONFIRMATION NO.
10/600,206	06/19/2003	Darko Segota	11023.5	9031
21999 KIRTON AND	7590 07/18/2007 MCCONKIE		EXAM	INER
60 EAST SOU	60 EAST SOUTH TEMPLE,		LEE, BENJAMIN P	
SUITE 1800 SALT LAKE (CITY, UT 84111		ART UNIT PAPER NUMBER	
	,		3641	
			MAIL DATE	DELIVERY MODE
			07/18/2007	PAPER

Please find below and/or attached an Office communication concerning this application or proceeding.

The time period for reply, if any, is set in the attached communication.

*****		Application No.	Applicant(s)			
Office Action Summary		10/600,206	SEGOTA ET AL.			
		Examiner	Art Unit			
		Benjamin P. Lee	3641			
Period f	The MAILING DATE of this communication app or Reply	pears on the cover sheet with the	correspondence address			
WHI0 - External after af	HORTENED STATUTORY PERIOD FOR REPLY CHEVER IS LONGER, FROM THE MAILING DATE on the may be available under the provisions of 37 CFR 1.13 or SIX (6) MONTHS from the mailing date of this communication. O period for reply is specified above, the maximum statutory period were to reply within the set or extended period for reply will, by statute, reply received by the Office later than three months after the mailing ned patent term adjustment. See 37 CFR 1.704(b).	ATE OF THIS COMMUNICATIO 36(a). In no event, however, may a reply be ti vill apply and will expire SIX (6) MONTHS fron cause the application to become ABANDON.	N. mely filed n the mailing date of this communication. ED (35 U.S.C. § 133).			
Status						
1)🖂	Responsive to communication(s) filed on 07 M	ay 2007.				
2a)⊠	This action is FINAL . 2b) This action is non-final.					
3)□	The state of the s					
	closed in accordance with the practice under E	x parte Quayle, 1935 C.D. 11, 4	53 O.G. 213.			
Disposit	ion of Claims					
4)⊠	Claim(s) 1-25 is/are pending in the application.					
,	4a) Of the above claim(s) 6 and 20-24 is/are with					
5)□	Claim(s) is/are allowed.					
6)⊠	Claim(s) <u>1-5, 7-19 and 25</u> is/are rejected.					
	Claim(s) is/are objected to.	·				
8)	Claim(s) are subject to restriction and/or	r election requirement.				
ຸ Applicat	ion Papers					
9)□	The specification is objected to by the Examiner	r				
	The drawing(s) filed on is/are: a) acce		Examiner			
,—	Applicant may not request that any objection to the					
	Replacement drawing sheet(s) including the correcti	- · · · · · · · · · · · · · · · · · · ·	` .			
11)	The oath or declaration is objected to by the Ex		- ·			
Priority ı	under 35 U.S.C. § 119					
	Acknowledgment is made of a claim for foreign	priority under 35 H S C & 110/a	\.(d) or (f)			
	☐ All b)☐ Some * c)☐ None of:	priority under 00 0.0.0. § 115(a)-(a) or (i).			
,	1. Certified copies of the priority documents	s have been received.				
	2. Certified copies of the priority documents have been received in Application No					
	3. Copies of the certified copies of the prior					
	application from the International Bureau		· ·			
* \$	See the attached detailed Office action for a list of	of the certified copies not receive	ed.			
Attachmen	nt(s)					
	ce of References Cited (PTO-892)	4) Interview Summary				
	ce of Draftsperson's Patent Drawing Review (PTO-948)	Paper No(s)/Mail D 5) Notice of Informal F	ate			
inton رے رد Pape	mation Disclosure Statement(s) (PTO/SB/08) r No(s)/Mail Date	6) Other:	алент Аррисацоп			

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DETAILED ACTION

1. Applicant has amended claims 1, 9, 11 and 18 and canceled claims 26-37.

Response to Arguments

2. Applicant's arguments with respect to claims 1-25 have been considered but are moot in view of the new ground(s) of rejection. With respect to Applicant's arguments regarding the combination of Wells et al and Smith et al, Examiner respectfully assert that Smith et al disclose an "offset" that constitutes a "pressure recovery drop" since the offset induces flow separation. Note that Smith et al disclose that the "flap" is capable of being "bowed" to form a negative camber and as seen in Smith et al figure 2, the negative camber in conjunction with the "offset" create a "pressure recovery drop". Further, Examiner respectfully maintains that the rejection under 35 U.S.C. 103(a) of claims 9, 10 and 12 was intended to affirm that it would have been obvious to make the "fluid flow regulator" dynamic or movable as disclosed by Smith et al, inherently providing the capabilities of adjustment and/or oscillation to the fluid flow regulator.

Claim Rejections - 35 USC § 103

The following is a quotation of 35 U.S.C. 103(a) which forms the basis for all obviousness rejections set forth in this Office action:

(a) A patent may not be obtained though the invention is not identically disclosed or described as set forth in section 102 of this title, if the differences between the subject matter sought to be patented and the prior art are such that the subject matter as a whole would have been obvious at the time the invention was made to a person having ordinary skill in the art to which said subject matter pertains. Patentability shall not be negatived by the manner in which the invention was made.

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The factual inquiries set forth in *Graham* v. *John Deere Co.*, 383 U.S. 1, 148 USPQ 459 (1966), that are applied for establishing a background for determining obviousness under 35 U.S.C. 103(a) are summarized as follows:

- 1. Determining the scope and contents of the prior art.
- 2. Ascertaining the differences between the prior art and the claims at issue.
- 3. Resolving the level of ordinary skill in the pertinent art.
- Considering objective evidence present in the application indicating obviousness or nonobviousness.
- 3. Claims 1-5, 7, 8, 13, 15, 16, 18 and 19 are rejected under 35 U.S.C. 103(a) as being unpatentable over Wells et al. (U.S. Patent 5505405) in view of Falco et al. (U.S. Patent 5133519).
- 4. In regards to claims 1 and 18, Wells et al disclose a fuselage comprising the following:
 - a frontal fuselage portion that leads through a fluid (col. 3, lines 55-59).
 Note that Wells et al disclose a "frontal portion of the fuselage" which inherently "leads through a fluid (air);
 - an outer fuselage surface relating with said frontal fuselage portion that
 receives fluid flow thereon (see Wells et al fig. 3 following);
 - c. at least one fluid flow regulator featured and operable with said outer fuselage surface and extending at least a partial distance around said fuselage (items 20 of Wells et al fig. 3 following and col. 4, lines 63-67);
 - d. fluid flow regulator comprising a leading surface and a trailing surface (see

Wells et al fig. 3 following);

- e. a pressure recovery drop extending a pre-determined distance between said leading and trailing edges/surfaces to form a down step, said pressure recovery drop comprising at least one drop face of a calculated distance (col. 3, lines 30-35), said fluid flow regulator functioning to regulate existing pressure gradients along said fuselage/surface subject to external flow of fluid to optimize and equalize said fluid flow and to reduce the separation potential of said fluid (see Wells et al fig. 3 and col. 3, lines 30-54);
- f. a sub-atmospheric barrier generated at the base of said drop face as said fluid encounters and flows over said pressure recovery drop, said sub-atmospheric barrier comprising a low pressure area of fluid molecules having decreased kinetic energy that serve as a cushion between said higher kinetic energy fluid molecules in said fluid and the molecules at said outer fuselage surface to facilitate laminar flow and assist in the reduction of the separation potential of said fluid (col. 3, lines 40-45). Note that the "step" disclosed by Wells et al creates a "low pressure area" and thus helps to reduce the separation potential of the fluid;
- g. a trailing edge that defines and extends from the base of said pressure recovery drop that provides a trailing flow boundary for said fluid (see Wells et al fig. 3 following). Note that the "trailing surface" extends from

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the "pressure recovery drop" and inherently provides a trailing flow "boundary" for the fluid;

Wells et al fail to explicitly disclose that the pressure recovery drop is orthogonal. However, Falco et al disclose a drag reducing device incorporating orthogonal pressure recovery drops (see Falco et al fig. 1 following). It would have been obvious to one of ordinary skill in the art at the time of Applicant's invention to incorporate orthogonal pressure recovery drops as disclosed by Falco et al, since orthogonal drops induce the greatest flow separation.

- 2. In regards to claim 2, Wells et al disclose a pressure recovery drop is positioned at or proximate an optimal pressure recovery point defined as the location(s) about said surface at which there is an imbalanced or unequal pressure gradient forward and aft of said fluid, thus creating an adverse pressure about said fuselage, which adverse pressure gradient induces friction and pressure drag that ultimately increases the separation potential of said fluid (col. 28-35). Note that Wells et al disclose positioning the "steps" at a point where the nose meets the fuselage and/or along the length of the fuselage.
- 3. In regards to claim 3, Wells et al disclose that the pressure recovery drop is oriented substantially perpendicular to the direction of flow of said fluid (see Wells et al fig. 3 following).

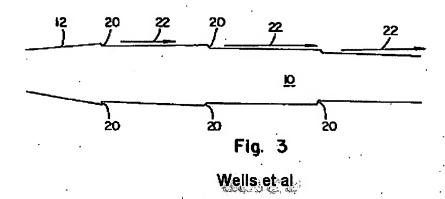
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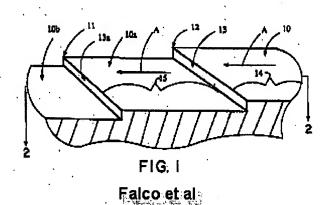
4. In regards to claim 4, Wells et al disclose that the pressure recovery drop comprises a linear formation (see Wells et al fig. 3 following).

- 5. In regards to claim 5, Wells et al disclose that the fluid flow regulator extends annularly around said fuselage (col. 4, lines 63-67).
- 6. In regards to claim 7, Wells et al disclose that the pressure recovery drop extends about only a portion of said outer fuselage surface (col. 3, lines 55-59). Note that Wells et al disclose that the "steps" extend longitudinally along the "forward portion" (only a portion) of the fuselage.
- 7. In regards to claim 8, Wells et al disclose that the outer fuselage surface features a plurality of fluid flow regulators that function together to regulate, influence, and control fluid flow and its properties and characteristics across said outer fuselage surface (see Wells et al fig. 3 following and col. 3, lines 39-45 and 55-59).
- 8. In regards to claim 13, Wells et al disclose that the fluid flow regulator is integrally formed with said outer fuselage surface (see Wells et al fig. 3 following).
- 9. In regards to claim 15, Wells et al disclose that the pressure recovery drop comprises a plurality of drop faces to magnify the influence of fluid flow regulator on said fluid (see Wells et al fig. 3 following).

10. In regards to claim 16, Wells et al disclose that the fuselage comprises a fuselage of an aircraft (col. 1, lines 5-12).

11. In regards to claim 19, Wells et al disclose that the moving body comprises the fuselage of an airplane or other similar aircraft (col. 3, lines 55-59).



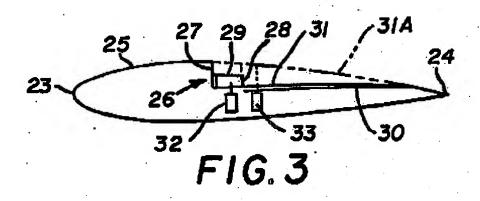


12. Claims 9, 10, 12, 17 and 25 are rejected under 35 U.S.C. 103(a) as being unpatentable over Wells et al. (U.S. Patent 5505405) and Falco et al. (U.S. Patent

5133519) as applied to claims 1 and 18 above, and further in view of Smith et al. (U.S. Patent 4890803).

13. In regards to claims 9, 10 and 12, Wells et al and Falco et al fail to disclose that the fluid flow regulator is a dynamic fluid flow regulator capable of adjusting, on demand, with varying design constraints, flow characteristics, environmental conditions, and operational situations pertaining to said fluid, said object, and any combination of these. However, Smith et al disclose a "fluid flow regulator" (item 26 of Smith et al fig. 3 following) that is "movable" to manipulate flow characteristics (col. 3, lines 42-46 and 57-65) and is inherently capable of "adjustment" to meet any of Applicant's stated conditions or situations. Note that this is equivalent to Applicant's adjustable "pressure recovery drop". It would have been obvious to one of ordinary skill in the art at the time of applicant's invention to incorporate a "movable" (adjustable) "fluid flow regulator" (pressure recovery drop) into/onto a fluid flow surface as disclosed by Smith et al, so that an operator can control the fluid flow dynamics and thus the lift generation of the fluid flow surface (body) is capable of manipulation according to the angle at which the fluid flow surface interfaces the direction of fluid flow (angle of attack). Note that based on the common definition of oscillate, "to move repeatedly from side to side or up and down between to points", presented by the Cambridge Dictionary of American English (www.dictionary.cambridge.org), the "movable pressure recovery drop" as stated by Applicant in claim 10, is inherently capable of "repeated [movement] between two points" and therefore Applicant's claim 12 is equivalent to claim 10.

14. In regards to claims 17 and 25, Wells et al fail to explicitly disclose that the pressure recovery drop comprises an orthogonal design. However, Smith et al disclose a "drop" that is at a right angle to the fluid flow surface (see Smith et al fig. 3 following). It would have been obvious to one of ordinary skill in the art at the time of applicant's invention to require that the "drop" is at a right angle to the flow surface as disclosed by Smith et al to achieve the most significant pressure drop.



Smith et al

- 15. Claims 11 and 14 are rejected under 35 U.S.C. 103(a) as being unpatentable over Wells et al. (U.S. Patent 5505405) and Falco et al. (U.S. Patent 5133519) as applied to claim 1 above, and further in view of Fronek et al. (U.S. Patent 5848769).
- 16. In regards to claim 11, Wells et al and Falco et al fail to disclose that the fluid flow regulator is capable of being repositioned in any direction about a surface. However,

Fronek et al disclose a removable "fluid flow regulator" (drag reduction article) that is inherently capable of being repositioned in any direction on a surface (col. 1, lines 10-35 and col. 5, lines 52-62). It would have been obvious to one of ordinary skill in the art at the time of Applicant's invention to make the fluid flow regulator "removable" as disclosed by Fronek et al, so that the fluid flow regulator can be replaced after damage.

17. In regards to claim 14, Wells et al and Falco et al fail to disclose that the fluid flow regulator that is removably attachable to allow said existing outer fuselage surface to feature one or more fluid flow regulators. However, Fronek et al disclose an "an article applied to surfaces to reduce the drag caused by fluids flowing across the surface" that is capable of being removeably attached to that surface (col. 1, lines 21-25). It would have been obvious to one of ordinary skill in the art at the time of applicant's invention to provide a "removable" fluid flow regulator as disclosed by Fronek et al, so that the "fluid flow regulator" for routine maintenance or to be replaced.

Summary/Conclusion

18. Claims 1-5, 7-19 and 25 are rejected.

THIS ACTION IS MADE FINAL. Applicant is reminded of the extension of time policy as set forth in 37 CFR 1.136(a).

A shortened statutory period for reply to this final action is set to expire THREE MONTHS from the mailing date of this action. In the event a first reply is filed within TWO MONTHS of the mailing date of this final action and the advisory action is not mailed until after the end of the THREE-MONTH shortened statutory period, then the shortened statutory period will expire on the date the advisory action is mailed, and any

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extension fee pursuant to 37 CFR 1.136(a) will be calculated from the mailing date of the advisory action. In no event, however, will the statutory period for reply expire later than SIX MONTHS from the mailing date of this final action.

Any inquiry concerning this communication or earlier communications from the examiner should be directed to Benjamin P. Lee whose telephone number is 571-272-8968. The examiner can normally be reached between the hours of 8:30am and 5:00pm on Monday through Friday.

If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Michael Carone can be reached on 571-272-6873. The fax phone number for the organization where this application or proceeding is assigned is 571-273-8300. Information regarding the status of an application may be obtained from the Patent Application Information Retrieval (PAIR) system. Status information for published applications may be obtained from either Private PAIR or Public PAIR. Status information for unpublished applications is available through Private PAIR only. For more information about the PAIR system, see http://pair-direct.uspto.gov. Should you have questions on access to the Private PAIR system, contact the Electronic Business Center (EBC) at 866-217-9197 (toll-free).

